

Impact of Gasification Fuels on Gas Turbine Operation

by

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Gas Turbine Plant Configurations



Integrated Gasification Combined Cycle with Polygeneration



Gasifier Types



Some IGCC Examples – US and Europe

Wabash R	iver, Indiana		Puertolleano, S	pain	
Tampa, Florida	Buggenum, Ti	he Netherlands			
Project / Location	Gasification Technology	Gas Turbine	Capacity (MWe)	Startup Year	
Wabash River / Indiana, USA	DOW-ConocoPhillips	GE-7FA	192	1995/2000	
Tampa Electric / Florida, USA	Tampa Electric / Florida, USAGE-Texaco		250	1996/2001	
Nuon Power, Demkolec / Buggenum, The Netherlands	Shell	Siemens – SGT5-2000E	253	1993/1998	
ELCOGAS / Puertollano, Spain	Uhde Prenflo	Siemens – SGT5-4000F	317	1996	

(Ref: Dennis, et al, 2007)



Some IGCC Examples – Canada





Project / Location	Gasification Technology	Gas Turbine	Capacity (MW)	Startup Year
OPTI-Nexen / Long Lake, AB	Shell	GE-7EA	170 (120/50)	2006
EPCOR Utilities / Genesee, AB	Siemens	Siemens	270	2015



Why IGCC with Polygeneration

- Lower CO₂ by 60~80%
- Lower air pollutants up to 95%
- Applicability to multiple sector
- Energy security through multiple feedstock
- Fuel flexibility through reduced reliance on natural gas and oil
- Simultaneous production of High-value by-products



NRC-CNRC



(Ref: EPCOR, 2008)

Gas Turbine Fuels – Conventional & Alternative



- Much tighter control (regulations) requirement on fuel specification in aviation gas turbines
- Requirement for industrial gas turbine – burn anything



LHV (MJ/kg)

(Ref: Wisniewski & Handelsman 2010)



Gas Turbine Fuels' Composition – Siemens' Experience



Gas Turbine Fuels' Properties – GE's Experience



	Main Constituents	LHV (MJ/m ³)		U/L Flammability Ratio	
		Min.	Max.	Min.	Max.
Natural Gas	CH_4 , C_2H_6	31.971	47.957	2.20	3.00
LPG	$C_{3}H_{8}, C_{4}H_{10}$	91.917	127.885	4.00	5.00
Air Blown Syngas	H_2 , CO, N_2 , H_2O , CO ₂	5.195	7.993	2.40	5.40
Oxygen Blown Syngas	H_2 , CO, H_2O , CO ₂	7,993	15.9 <u>8</u> 6	6.00	12.00
Blast Furnace Gas	H_2 , CO, N_2 , H_2O , CO ₂	2.997	4.996	1.50	3.00
Refinery Off-gas	$H_2, C_2H_6, C_3H_8, C_4H_{10}, C_2H_4, C_3H_6$	11.989	63.942	3.00	18.00
Coke Oven Gas	H_2 , CO, N_2 , H_2 O, CO ₂	11.989	19.982	6.00	8.00



Syngas Related Issues – Composition Variations

- Gasifier type
 - Oxygen vs. air blown
 - Dry vs. slurry fed
- Process temperature
- Feed rate
- Amount of Oxygen
- H:C ratio in feedstock

Composition (Volume %)	Coal-Gas	Bio-Gas	Natural Gas
Hydrogen (H ₂)	14.0%	18.0%	
Carbon Monoxide (CO)	27.0%	24.0%	
Carbon Dioxide (CO ₂)	4.5%	6.0%	
Oxygen (O ₂)	0.6%	0.4%	
Methane (CH ₄)	3.0%	3.0%	90.0%
Nitrogen (N ₂)	50.9%	48.6%	5.0%
Ethane (C ₂ H ₆)			5.0%
HHV (kJ/m ³)	6,417	5,315	39,450

Feedstock Variation

Process Variation

Composition (Volume %)	Min.	Max.	Avg.
Hydrogen (H ₂)	8.6	61.9	31.0
Carbon Monoxide (CO)	22.3	55.4	37.2
Carbon Dioxide (CO ₂)	1.6	30	12
Methane (CH ₄)	0	8.2	2.2
Nitrogen (N ₂) + Argon (Ar)	0.2	49.3	12.2
Water (H ₂ O)	0.1	39.8	7.8
Hydrogen/Carbon Monoxide Ratio	0.33	0.8	0.86



Fuel Constituents – Characteristic Values

		I HV	I HV	Flammabi (Vo	lity Limits I. %)	Autoignition	Laminar Flame Speed (cm/s)
		(MJ/m ³)	(MJ/kg)	Lean	U/L Ratio	()	
Methane	CH₄	36.447	50.048	5.00	3.00	537	44.8
Ethane	C_2H_6	64.862	47.511	3.00	4.13	472	47.6
Propane	C_3H_8	92.836	46.330	2.10	4.52	450	46.4
Butane	C_4H_{10}	120.651	45.725	1.80	4.67	462	44.9
Pentane	C_5H_{12}	148.586	45.343	1.40	5.57	284	43
Hexane	C_6H_{14}	176.441	44.925	1.20 🕈	6.17	225	
Carbon Monoxide	CO	12.828	10.113	12.50	5.92	609	52
Hydrogen	H_2	10.990	120.071	4.00	18.75	400	_325_



Fuel Flexibility Spread



Low reactivity fuels

(Ref: Wisniewski & Handelsman 2010)



Fuel Flexibility Challenge

- How does/can non-conventional (high vs. low reactivity) fuels affect gas turbine operation?
 - Combustion
 - Turbomachinery
 - Emissions
 - Hot gas path components
 - Maintenance
- Decision to utilize alternative fuels depends on these effects and the associated economics



Turbomachinery Issues

- Variation in enthalpy drop in the expansion
- Variation of the flow rate at turbine inlet and the effect on turbine compressor matching
- Variation in heat transfer coefficient on the turbine blades, affecting blade cooling performance





Important Combustor Performance Parameters

- Wide operability
 - Blow-off limits
 - Flashback and auto-ignition limits
 - Static and dynamic stability (spatial and temporal flame anchoring)
- Low emission
- Good turndown
- Durability



Fuel Composition Issues – Flame Blowoff



Conditions: U_0 =60 m/s, T=460K, P=4.4atm,

Fuel Composition Issues – Flame Flashback

Multiple flashback mechanisms

- In boundary layer
- In core flow
- Strong acoustic pulsations lead to nearly reverse flow
- Combustion induced vortex breakdown

Different fuel properties influence these mechanisms differently

- Strong dependence of turbulent flame speed on fuel composition
- Hydrogen influence on flashback



Fuel Composition Issues – Flame and
Combustion Stability

- Fuel composition variations influence
 - Flame shape
 - Flame standoff location
- Alteration in flame shape and location can worsen or improve combustor dynamics via $\tau_{convect}$





Fuel Composition Issues – Emissions

- Strongly dependant on composition
- Reactive fuel blends having high H₂ or C₂+ compositions
 - Increase NOx formation
 - Decrease CO formation at part load
- Fuels having high inert constituents
 - Reduce NOx formation
 - Increase concentration of CO and UHC in exhaust



Syngas Emissions

- Strongly dependant on composition
- In general syngas produce lower emissions for combined cycles
- VOC emissions low
- SOx emissions low
- CO emissions
 - Unburned syngas CO from insufficient mixing and equivalence ratio lower than ignition range
 - Incomplete combustion of HC contents
- NOx emissions
 - Thermally generated: Increase with increase in H2 contents due to higher firing temperatures. Decrease with increase in H2 contents due to leaner combustion potentials
 - Flame-generated: Increase with increase in H2 contents due to higher flame temperatures
 - Fuel-bound: Increase if ammonia not removed prior to combustion.
 Decrease if burned rich.
 - Increase with increase in CO:H2 ratio



Dry Low NOx Operation within Emissions & Dynamics Limits



Effects on Hardware Changeability & Durability

- Increased fuel reactivity causes thermal distress to premixer and hot gas-path components due to:
 - Higher flame temperature and flashback propensity
 - Susceptibility to high temperature thermoacoustic pressure oscillations
- High reactivity fuels require
 - Alternate fuel as well as purging system for starting and shutdown
- Reduced fuel reactivity due to addition of Inerts require
 - Larger sized injectors to compensate for higher fuel flow rate requirement
- Reduced fuel reactivity cause hardware distress due to
 - Low temperature combustion dynamics
- Syngas use may cause increased component corrosion



Solutions to Fuel Flexibility Challenge – Rolls-Royce Experience

Series Staging

- Simple control system with only 4 fuel control valves
 - diffusion, primary and secondary (2) VSS
- Diffusion circuit allows for reliable starting
- Primary and secondary enables individual zone temperature control.
 - This gives flexibility for optimum emissions control.



Series Staging– Operational Mapping



(Ref: Rolls-Royce, 2011)



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