

2018 FALL WORKSHOP
Gas Turbine Energy Systems:
Clean and Reliable Energy on Demand
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GT Combustion and Emissions

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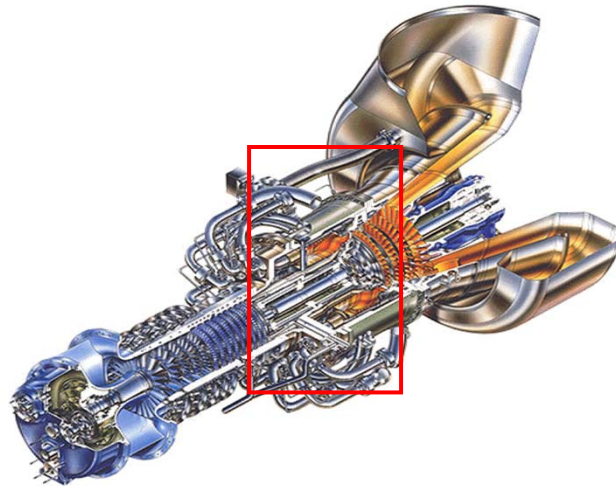
AGENDA

- Combustion
- Emissions
- Combustion System Description
- Fuels and Fuel Flexibility
- Summary

Caterpillar: Non-Confidential

Caterpillar: Confidential Green

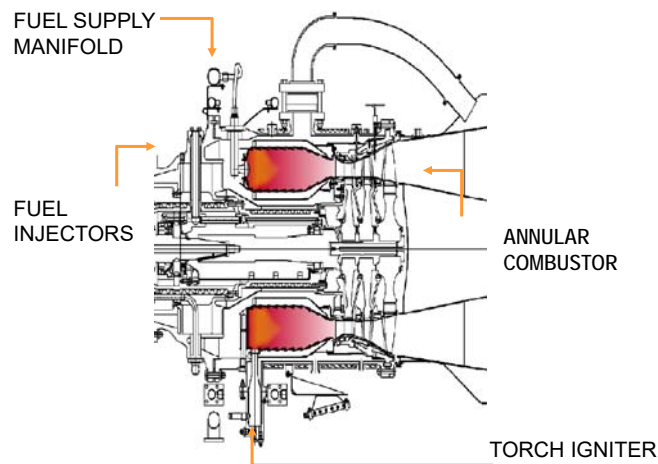
COMBUSTOR



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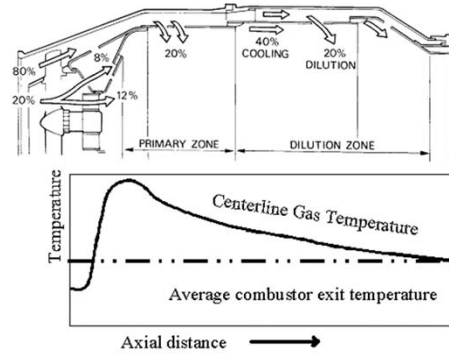
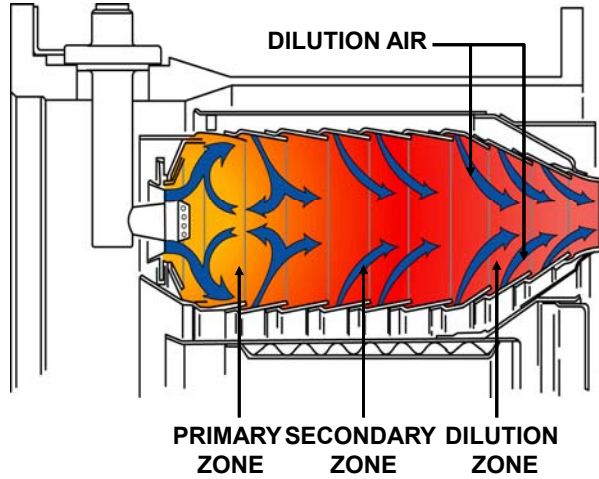
TYPICAL GAS TURBINE COMBUSTION SYSTEM



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Combustion Chamber



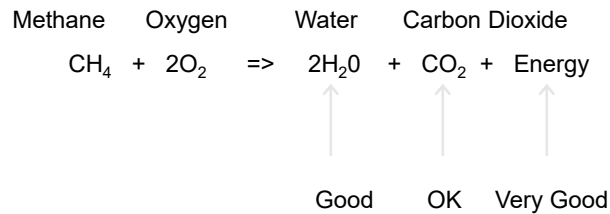
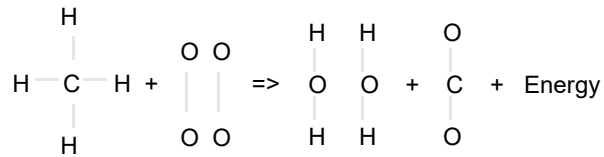
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Combustion

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Combustion Reactions



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971355-045

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Emissions

Air Is 78% Nitrogen (N₂), 21% Oxygen, Fuel is Hydrocarbons (C and H):

Incomplete Combustion:

- Unburned Hydrocarbons, Volatile Organic Compounds
- CO

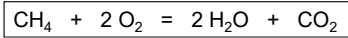
Combustion Byproducts:

- NO, NO₂

If the fuel contains Sulfur (Sulfur, H₂S, Mercaptanes), they will be completely oxidized

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Theoretical (Stoichiometric) Air-to-Fuel Ratio for Ideal Combustion



Molecular Weights:
Methane (CH₄) = 16.04
Oxygen (O₂) = 31.99

$$\left[\frac{\text{Oxygen}}{\text{Methane}} \right]_{\text{weight}} = \frac{2 \cdot 31.99}{16.04} = 3.99$$

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Theoretical (Stoichiometric) Air-to-Fuel Ratio for Ideal Combustion

Air Is Approximately 21% Oxygen

thus:

$$\left[\frac{\text{Air}}{\text{Fuel}} \right]_{\text{weight}} = \frac{3.99}{0.21} = 18.99$$

To Burn 1 kg of Methane We Need 18.99 kg of Air

Actual for Mars 100:

$$\left[\frac{\text{Air}}{\text{Fuel}} \right] = 65!$$

Because of Temperature Limitation.

Note: Actual Flame Front is about stoichiometric (in a conventional system)

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TYPES OF COMBUSTION FLAMES



- Diffusion Flame (*Conventional*)
 - Air, Fuel Injected into Combustor Separately
 - Burning at Interface (Flame Front), *Stoichiometric Fuel-to-Air Ratio*



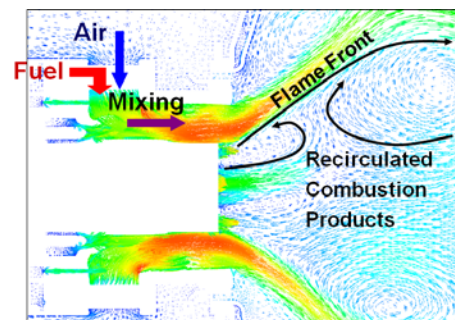
- Lean, Premixed Flame (*SoLoNOx, DLN, DLE*)
 - Air, Fuel *Premixed* in Fuel Injector
 - Burning at Lower Temperature, *Fuel-Lean Conditions*

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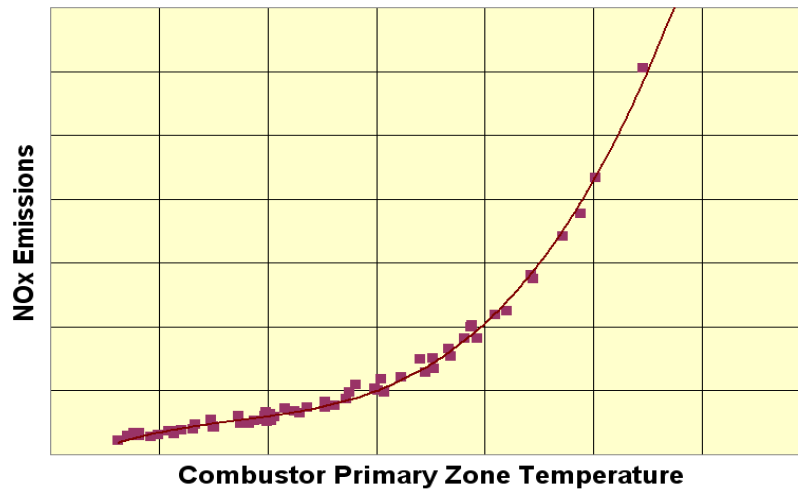
A Few Comments on NOx

- The best NOx is always achieved with perfect premixing of fuel and air.
- NOx emissions for well-mixed systems are independent of flameholder geometry
- A well-mixed flame is insensitive to pressure or inlet temperature. A diffusion flame is sensitive to both, leading to high pressure ratio engines having higher NOx than low pressure ratio machines. But a high pressure ratio DLE system should be able to achieve the same NOx as a low pressure ratio machine.
- NOx for a well-mixed system is independent of residence time. So a DLE system can have a larger combustor liner without increasing Nox to allow for complete combustion.



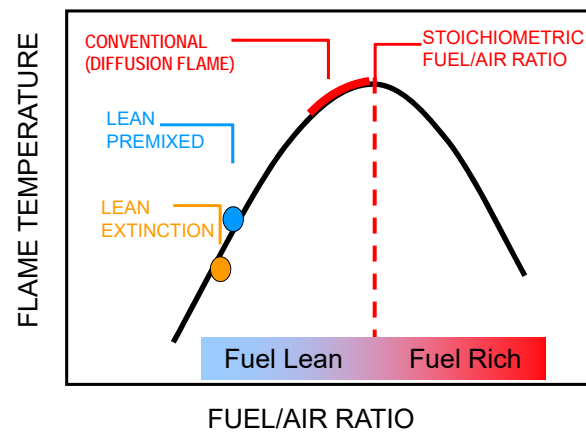
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Nox is a function of Flame Temperature



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STOICHIOMETRY EFFECT ON FLAME TEMPERATURE



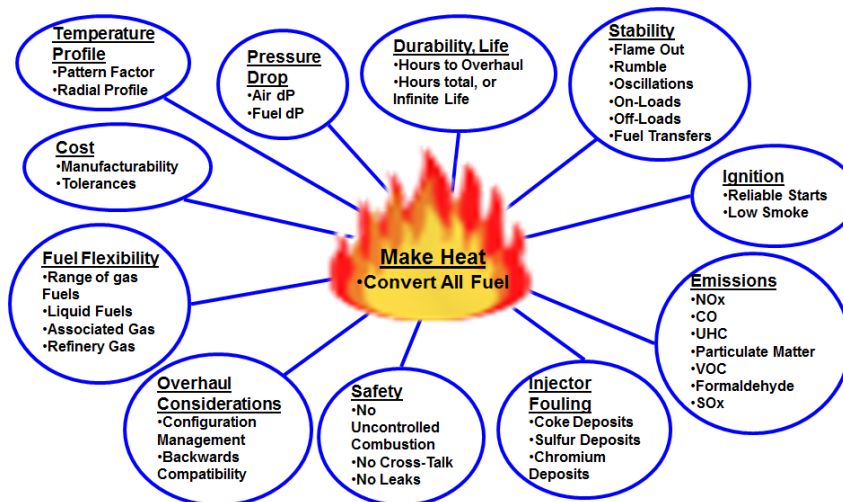
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Combustion Systems

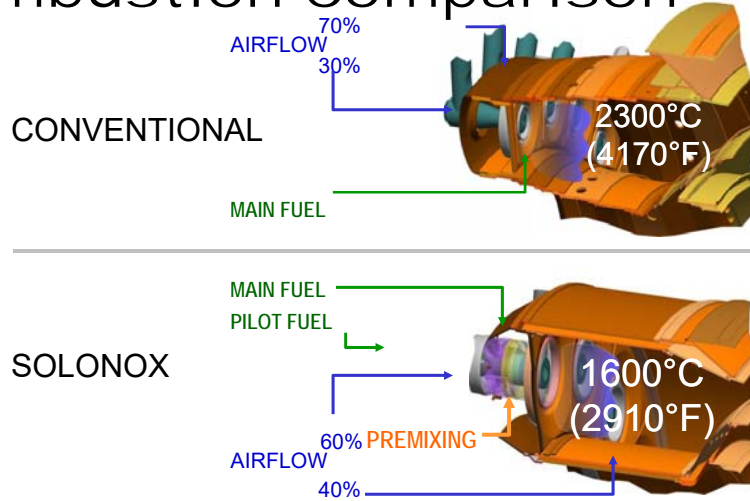
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Design Requirements and Considerations for a Gas Turbine Combustion System



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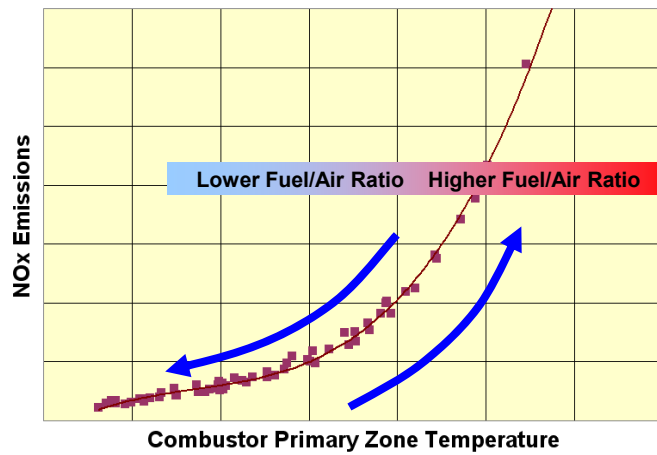
Combustion comparison



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NOX EMISSIONS CHARACTERISTIC



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COMBUSTION SYSTEM COMPONENTS

COMBUSTOR LINERS



Conventional SoLoNOx



FUEL INJECTORS



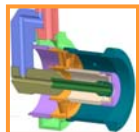
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FEATURES FOR IMPROVED EMISSIONS



ELECTRIC INLET GUIDE VANES OR BLEED VALVES



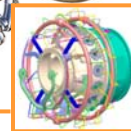
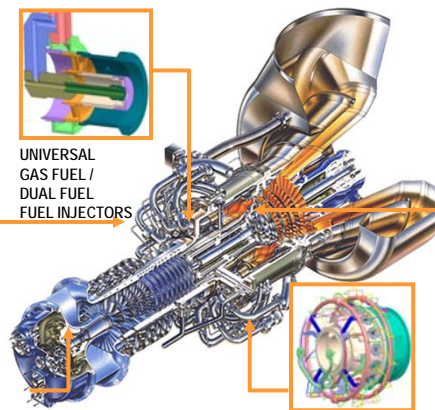
UNIVERSAL GAS FUEL / DUAL FUEL FUEL INJECTORS



AUGMENTED BACKSIDE COOLED (ABC) COMBUSTOR LINER



PARALLEL FUEL VALVE MODULE



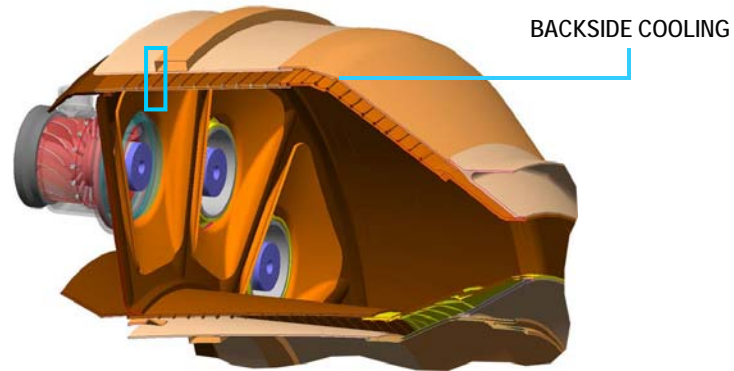
FUEL SYSTEM PLUMBING

- Fuel Flow Metering
- Closed Loop Pilot Logic
- Fully Integrated System

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AUGMENTED BACKSIDE COOLED - (abc) LINER TECHNOLOGY



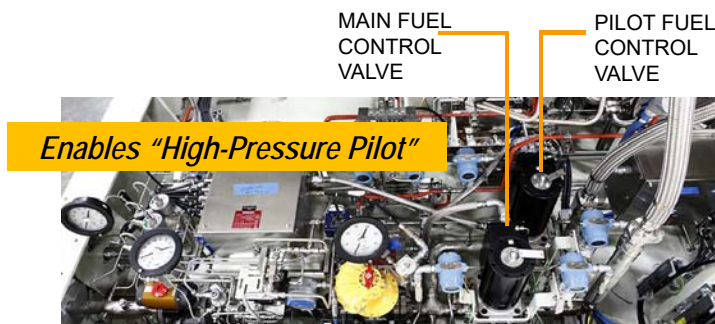
- No Film or Effusion Cooling in Primary Zone
- Rumble, CO Improved over Previous Generations

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FUEL SYSTEM IMPROVEMENTS

- Parallel Main and Pilot Fuel Valve Arrangement
- Improved % Pilot Control
 - Based on Main and Pilot Flow Measurements

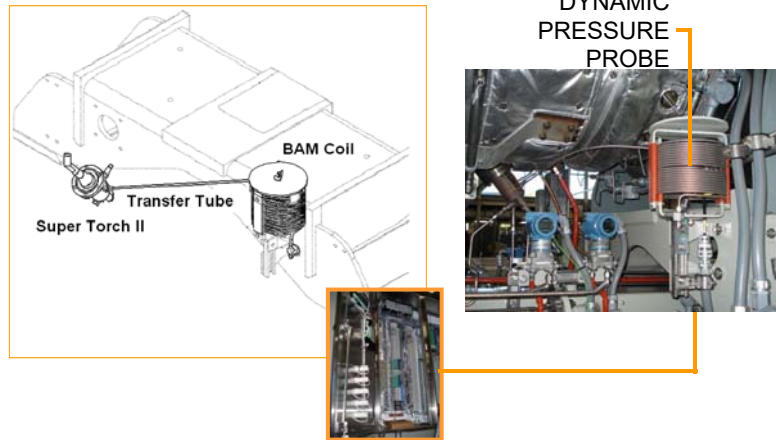


GAS ONLY FUEL MODULE

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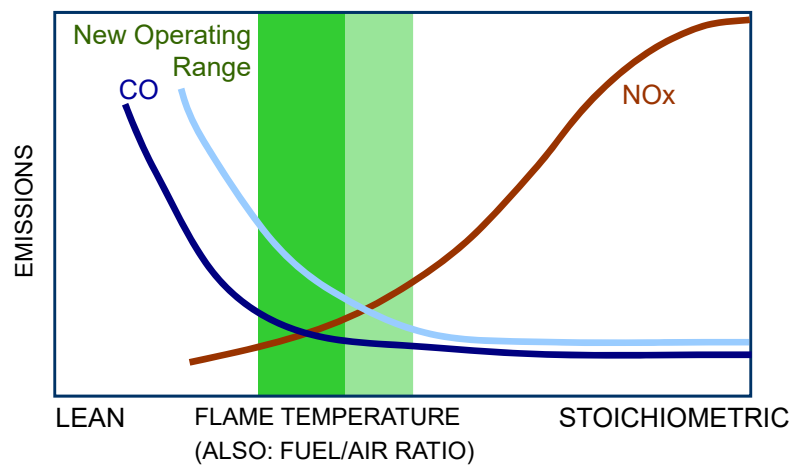
BURNER ACOUSTIC MONITOR – STANDARD



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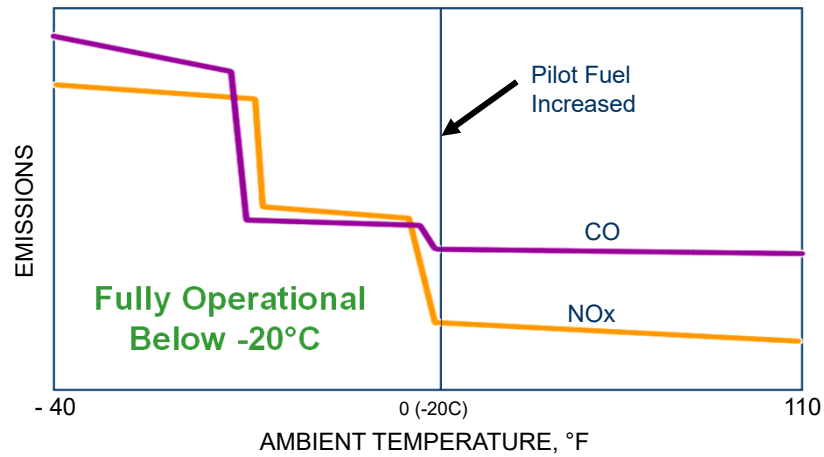
COMBUSTION SYSTEM EMISSIONS CHARACTERISTICS



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GENERAL TRENDS FOR NOX AND CO WITH AMBIENT TEMPERATURE



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GASEOUS FUEL TERMINOLOGY

Wobbe Index

$$WI = \frac{LHV}{\sqrt{SG}}$$

LHV = Lower Heating Value

SG = Specific Gravity

Corrected Wobbe Index

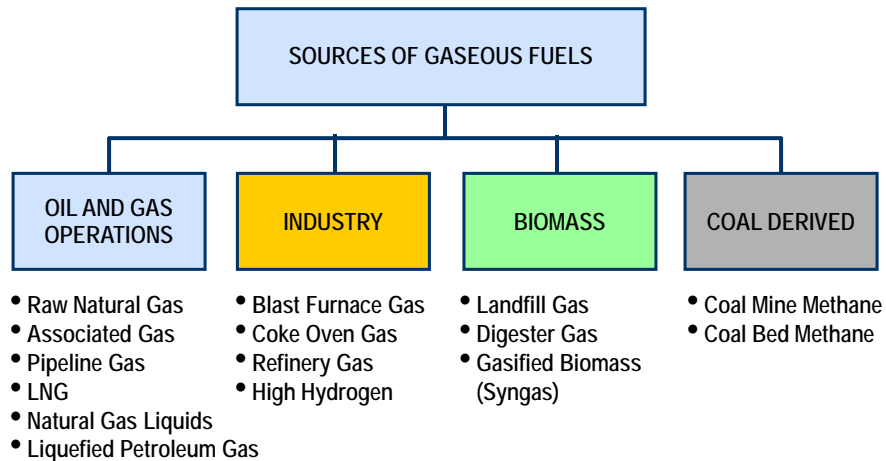
$$WI_{CORR} = WI \sqrt{\frac{60 + 459.67}{T_{fuel} + 459.67}}$$

U.S. Pipeline Natural Gas Ranges <i>Gas Processors Association (1998)</i>				
	Btu/scf		MJ/Nm ³	
	Minimum	Maximum	Minimum	Maximum
Higher Heating Value	950	1150	35.4	42.9
Lower Heating Value	856	1040	31.9	38.8
Wobbe Index	1085	1296	40.5	48.3

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GASEOUS FUELS SUITABLE FOR GAS TURBINES



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GASEOUS FUEL CONSIDERATIONS

Conventional

- Heating Value
 - Skid Edge Pressure
- Dew Point
 - Liquid Dropout
- Flammability Limits
- Adiabatic Flame Temperature
 - NOx Emissions
- Contaminants

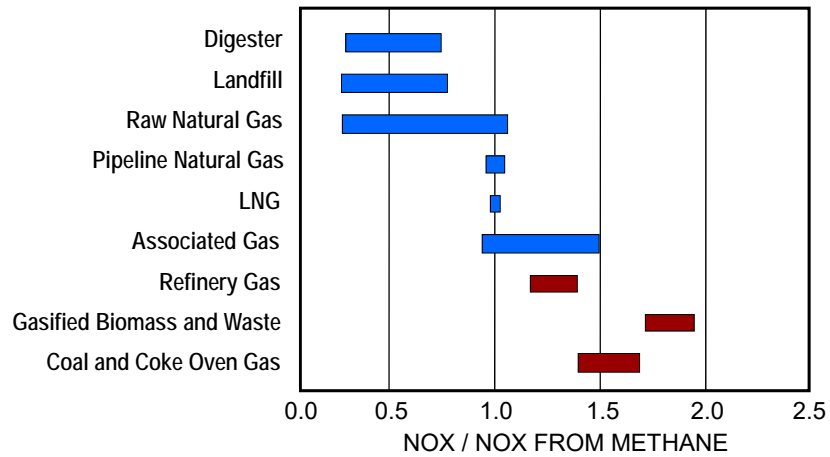
SoLoNOx

- Same as "Conventional"
- Plus:
 - Flame Speed (C4⁺+H₂+Alkenes)
 - Autoignition Delay Time
 - Emissions
 - Oscillations

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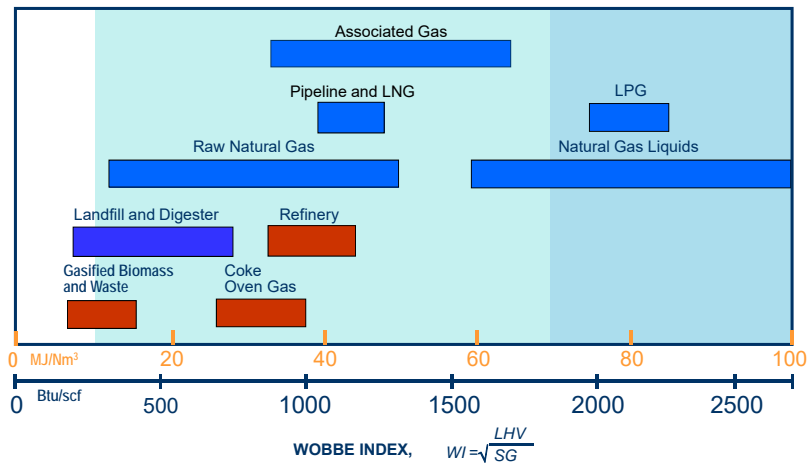
GAS NOX EMISSIONS RULE OF THUMB (CONVENTIONAL)



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GASEOUS FUEL RANGE - Conventional

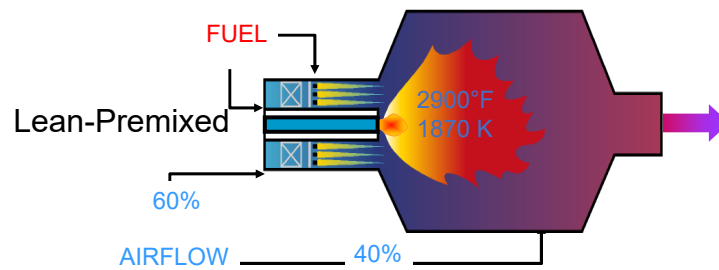


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Auto Ignition and Flashback

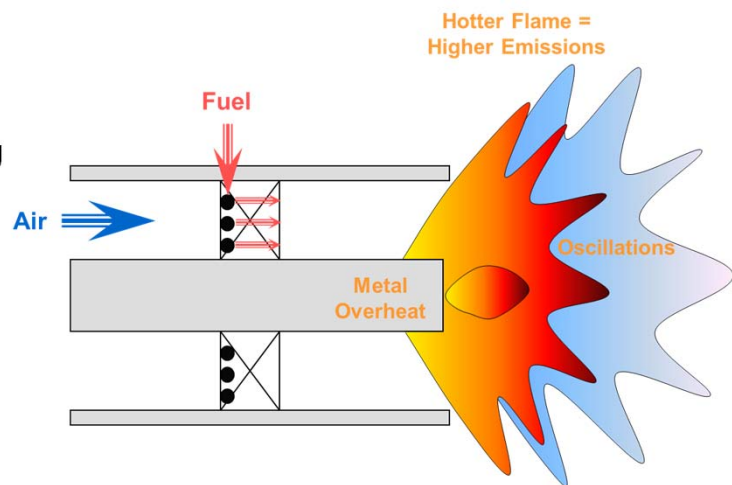
- Autoignition Delay Time
 - Longer for lean mixtures
 - Shorter for higher pressure and temperature
- Flashback
 - Flame Speed vs Bulk Velocity
 - Flame speed higher for Propane or Hydrogen



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Flame Instability

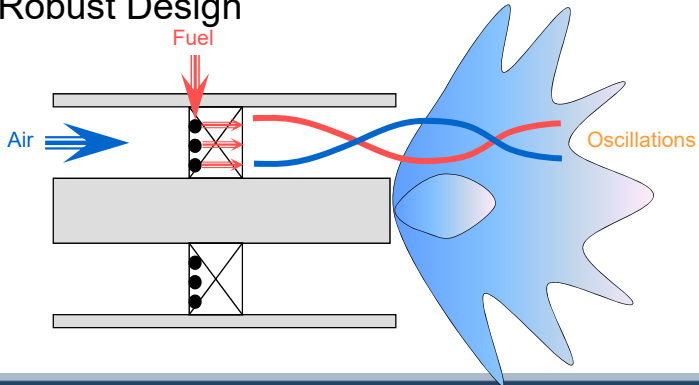
- Flame Movement
 - Durability Issues
 - Emissions Issues
 - Acoustic Coupling



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Combustion Acoustic Modeling

- Enhance Understanding of Oscillations
- Provide Accurate Predictions
- Allow for Up-Front Robust Design



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GASEOUS FUEL BLENDING FACILITIES



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SUMMARY

- SoLoNOx Significantly Lowers Emissions with Some Effect on Complexity
- Recent Advancements in SoLoNOx Technology
 - Improved Operability and Durability
 - Range of Conditions for Low Emissions
- Wide Range of Fuels Can Be Run in Conventional Combustion Systems
- Development and Qualification in Process to Expand Fuel Flexibility of SoLoNOx Systems

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